

Lecture 6

Parameterization / Specifying Rotations

Rot. 3×3 matrices, orthonormal

$[r_{i;j}]$ 9 elements

$R(a, b, c)$ ① Col. vec. are unit vec.
② " " " mutually orth.

③ $\det(R) = 1$ does not change
the # of ind. par.

\Rightarrow 3 ind. elements in the matrix

Several diff. conventions /way to rep.
these ind. elements.

1) Fixed angle representation $\{A\}$
We computed (axes)

$$R_z(\theta) \text{ in } \begin{array}{l} \text{1) around } \hat{x}_A \text{ by angle } \gamma \\ \text{Lecture 2. 2) " } \hat{y}_A \text{ " " } \beta \\ = \begin{pmatrix} \cos \theta & -\sin \theta & 0 \\ \sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{pmatrix} \end{array}$$

$$\text{similarly } R_{x'}(\theta), R_x(\theta) \quad R_{XYZ}(r, \beta, \kappa) = R_z(\kappa) R_y(\beta) R_x(r)$$

can be d. Given r, β, κ , we can compute
compute see below.

$$R_{XYZ}(r, \beta, \kappa)$$

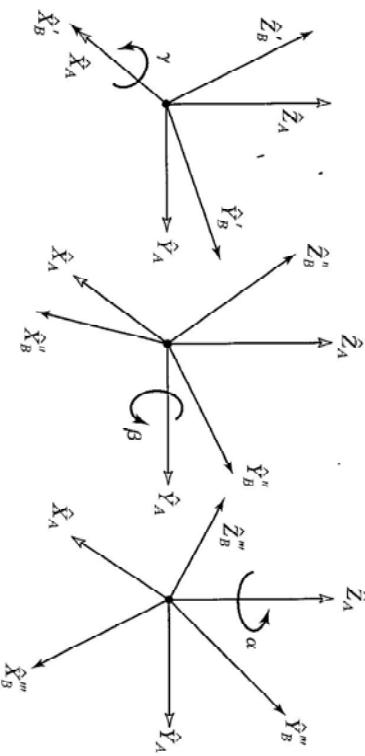


FIGURE 2.17: X–Y–Z fixed angles. Rotations are performed in the order $R_X(\gamma)$, $R_Y(\beta)$, $R_Z(\alpha)$.

The derivation of the equivalent rotation matrix, ${}^A R_{XYZ}(\gamma, \beta, \alpha)$, is straightforward, because all rotations occur about axes of the reference frame, that is,

$${}^A R_{XYZ}(\gamma, \beta, \alpha) = R_Z(\alpha)R_Y(\beta)R_X(\gamma)$$

$$= \begin{bmatrix} c\alpha & -s\alpha & 0 \\ s\alpha & c\alpha & 0 \\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} c\beta & 0 & s\beta \\ 0 & 1 & 0 \\ -s\beta & 0 & c\beta \end{bmatrix} \begin{bmatrix} 1 & 0 & 0 \\ 0 & c\gamma & -s\gamma \\ 0 & s\gamma & c\gamma \end{bmatrix}, \quad (2.63)$$

where $c\alpha$ is shorthand for $\cos \alpha$, $s\alpha$ for $\sin \alpha$, and so on. It is extremely important to understand the order of rotations used in (2.63). Thinking in terms of rotations as operators, we have applied the rotations (from the *right*) of $R_X(\gamma)$, then $R_Y(\beta)$, and then $R_Z(\alpha)$. Multiplying (2.63) out, we obtain

$${}^A R_{XYZ}(\gamma, \beta, \alpha) = \begin{bmatrix} cac\beta & cas\beta\gamma - sac\gamma & cas\beta\gamma + sac\gamma \\ sac\beta & sacs\beta\gamma + cac\gamma & sacs\beta\gamma - cas\gamma \\ -s\beta & c\beta s\gamma & c\beta c\gamma \end{bmatrix}. \quad (2.64)$$

Keep in mind that the definition given here specifies the order of the three rotations.

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If $\beta =$

Z-Y-

2) Euler Angle rep.: successive

rotations are carried out with respect to "current" frame and not original frame.

$$\{A\} \xrightarrow{\hat{z}_A, \alpha} \{B'\} \xrightarrow{\hat{y}_{B'}, \beta} \{B''\} \xrightarrow{\hat{x}_{B''}, \gamma} \{B'''\}$$

$$R_{Z'Y'X'}(\alpha, \beta, \gamma) = R_A^{\alpha} R_B^{\beta} R_C^{\gamma}$$
$$= R_Z^{\alpha} R_Y^{\beta} R_X^{\gamma}$$

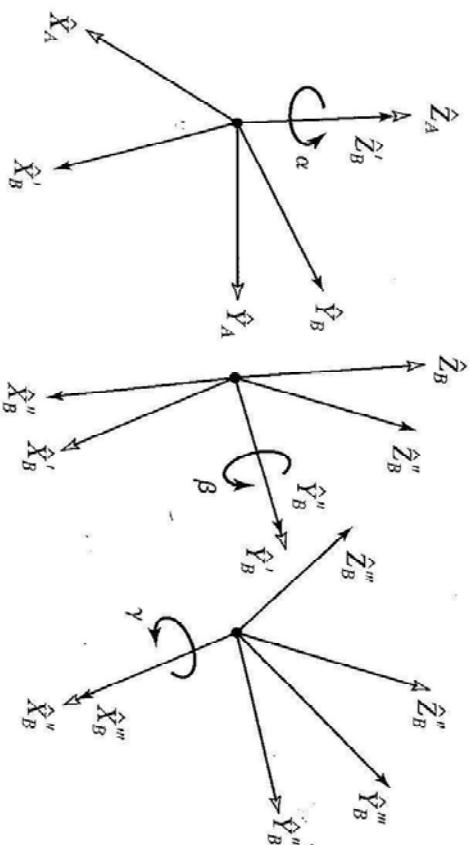


FIGURE 2.18: Z-Y-X Euler angles.

Another

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 Z_b

Rc

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frame.
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be used

Current axes rotation \rightarrow post. mult. of
matrices

fixed " " \rightarrow pre mult of
matrices.

why? the connection? latin?

easy

forward comp. \rightarrow Given α, β, γ , compute

$$\alpha, \beta, \gamma \xrightarrow{\text{f.}} R \quad R_{XYZ}(\alpha, \beta, \gamma)$$

inv.

inverse " " \rightarrow Given, R , compute
 α, β, γ .

Given $R_{z'y'x'}(\alpha, \beta, \gamma)$, Compute

Given $\begin{bmatrix} r_{11} & r_{12} & r_{13} \\ r_{21} & r_{22} & r_{23} \\ r_{31} & r_{32} & r_{33} \end{bmatrix}$, Solve for α, β, γ

Comparing element by element, you get 9 eqns.

highly non-linear, multiple variables (α, β, γ).

Tricks: look for eqns. from where you

can eliminate some variables using

trig. identification.

Always use $\text{Atom}_2(y, x)$, it involves the angle uniquely in all four quadrants.

by inspection,

$$\boxed{\tan z(y, x) = \theta} \quad g_{11} = c \sec \beta \quad \Rightarrow \quad c \beta = \pm \sqrt{\lambda_{11}^2 + \lambda_{12}^2 - 1}$$

$$\frac{dy}{dx} = \frac{1}{\sec \beta}$$

Now,

$$g_{31} = -s \beta \quad (2)$$

using (1) + (2), and assuming $|\beta| \leq 90^\circ$

$$\text{we get } \boxed{\beta} = \operatorname{atan}^2 \left(-g_{31} \right) \underbrace{\sqrt{\lambda_{11}^2 + \lambda_{12}^2}}_{\text{+ve sign}}$$

Now, we know β . Going back to λ_{11} & λ_{12} , we can divide by $c \beta$, to get $c \alpha$, $s \alpha$, respectively.
It follows,

$$\boxed{\alpha} = \text{Atan2}\left(\alpha_{z1}/c\beta, \alpha_{u1}/c\beta\right)$$

Assumes $\beta \neq 90^\circ$. More on it later.

by inspection,

$$\left. \begin{array}{l} \alpha_{32} = c\beta \wedge \gamma \\ \alpha_{33} = c\beta c \gamma \end{array} \right\} \Rightarrow \boxed{\gamma} = \text{Atan2}\left(\alpha_{32}/c\beta, \alpha_{33}/c\beta\right)$$

Done: we have solved for α, β, γ assuming $\beta \neq 90^\circ$.

What if $\beta = 90^\circ$: matrix form becomes

$$\begin{bmatrix} 0 & \rightarrow (x-\alpha) \\ 0 & c(x-\alpha) \\ -1 & -\rightarrow (x-\alpha) \\ 0 & \end{bmatrix}$$

\Rightarrow we can only solve for $(x-\alpha)!!$.

and not x, α individually. Also

\Rightarrow infinite nohn. all x, α such that $x-\alpha$ remains same!

Such situations are called "singularities"
and need to be explicitly taken into
account.

Why do they happen?: often axes line up!!
 $\{3 = q_0 \cdot \hat{x}''' \}$ (axis of third
rotation, by γ) lines up with $-\hat{z} = \hat{z}'$
no wr rot. of $(\gamma - \alpha)$ around $-\hat{z}$ axis.

3) Equivalent angle - axis rep.

$$R_k(\theta) \quad \begin{matrix} \text{given axis of rot.} \\ \text{is a unit vec. } \hat{k} \end{matrix}$$
$$\hat{k} = \begin{pmatrix} k_x \\ k_y \\ k_z \end{pmatrix} \quad \equiv \quad \theta = \text{amount of rotation.}$$

$$|\hat{k}| = 1$$

form of this matrix : ??

basic idea: Carry out successive rotations

to align \hat{k} with \hat{z} axis (or any principal axis),

carry out rot. by θ around \hat{z} , then rotate \hat{z} back to \hat{k} .

Convince your self (carry this out) that it
physically leads to same final orientation !!

See figure below : Note in fig :

$$k_x = g_x$$

$$k_y = g_y$$

$$k_3 = g_3$$

$$\Theta = \phi$$

$$\sin \alpha = \frac{r_y}{\sqrt{r_y^2 + r_z^2}} \quad \cos \alpha = \frac{r_z}{\sqrt{r_y^2 + r_z^2}}$$

$$\sin \beta = r_x \quad \cos \beta = \frac{\sqrt{r_y^2 + r_z^2}}{r_x}$$

Substituting into the above equation,

$$\mathbf{R}_{r,\phi} = \begin{bmatrix} r_x^2 V\phi + C\phi & r_x r_y V\phi - r_z S\phi & r_x r_z V\phi + r_y S\phi \\ r_x r_y V\phi + r_z S\phi & r_y^2 V\phi + C\phi & r_y r_z V\phi - r_x S\phi \\ r_x r_z V\phi - r_y S\phi & r_y r_z V\phi + r_x S\phi & r_z^2 V\phi + C\phi \end{bmatrix}$$

$$C\phi = \cos \phi$$

$$V\phi = 1 - \cos \phi$$

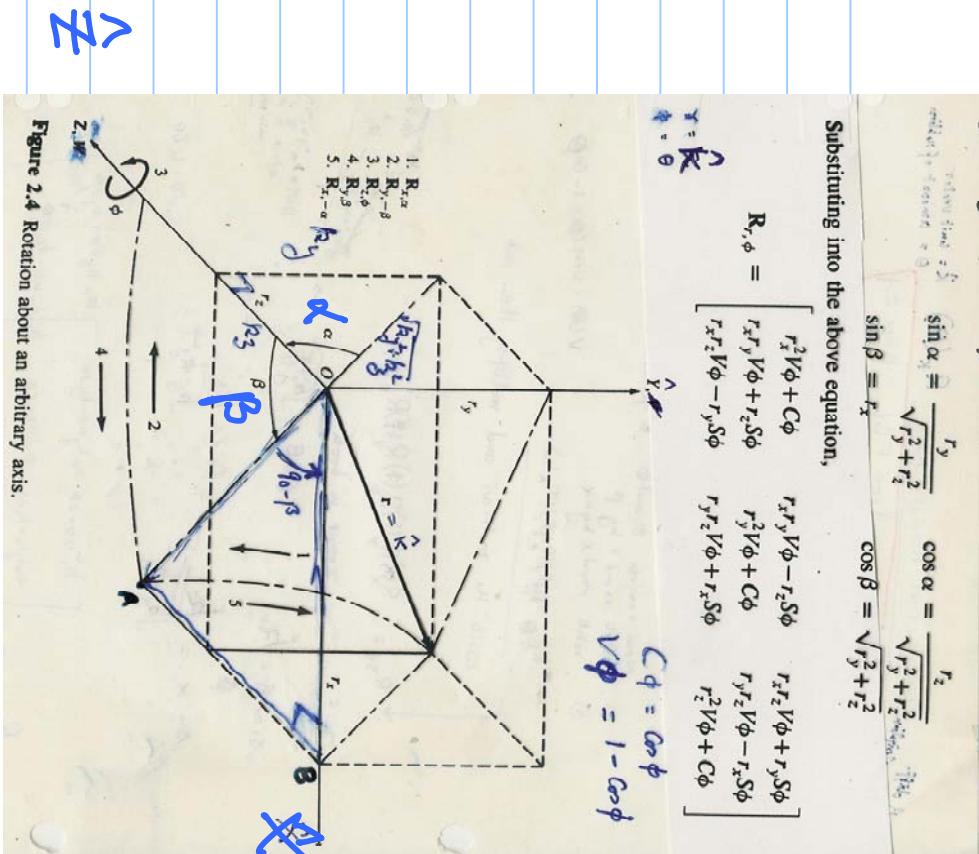


Figure 2.4 Rotation about an arbitrary axis.

Convince yourself that $R_{\hat{z}}(\theta)$ is

Equivalent to following rotations:

1) Rotate \hat{k} by α around \hat{x} axis. This brings \hat{k} to x - \hat{z} plane.

2) Rotate by $-\beta$ around \hat{y} axis.

This brings \hat{k} to align with \hat{z} .

actual
rotation

3) Rotate by θ around \hat{z} .

4) Reverse of 2 : rotate by β around \hat{y} - axis.

5) Reverse of 1 : rotate by $-\alpha$ around \hat{x} .

Since all rot. are around fixed axes of the same frame, pre-multiply. So, we get:

$$R_k^{\alpha}(\theta) = R_y^{(\alpha)} R_y^{(\beta)} R_z^{(\theta)} R_y^{(-\beta)} R_x^{(\alpha)}$$

α, β, θ are related to r_x, r_y as shown in Fig. by straight forward geometry. Multiplying all out, we get the result stated in text in Eqn. 2.80.