

The Greatest Engineering Race: Part 3

Cheap Access to Space

The New Race to Space & The Key to our Future in Space

By Glenn Chapman

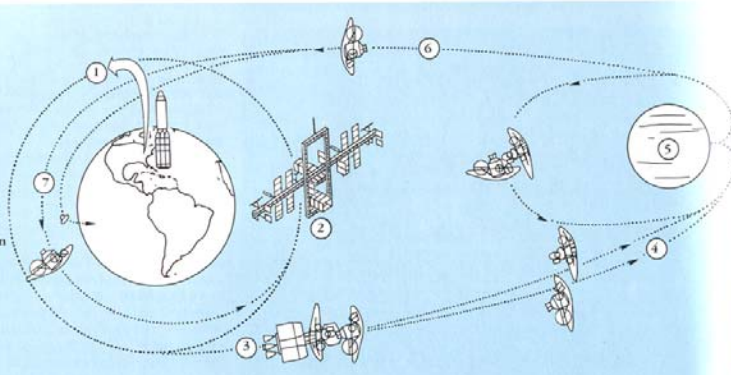


Why Go Into Space?

- Exploration of the Moon & Planets
- When you are in orbit you are half way to the planets
- Space Manufacturing in Low Earth Orbit
- Raw Materials: asteroids & planets
- Robot vs Human

A mission profile for early base emplacement on Mars.

1. Cargo and crew of the Mars-bound vehicle are prepared at Freedom "transport node".
2. Mars Transfer Vehicle, housing a crew of four, is mated with rest of vehicle.
3. Trans-Mars Injection Stage fires to send whole craft toward the Red Planet.
4. Mars Lander separates and arrives on Mars 7 days before Transfer Vehicle, which remains in Mars Orbit.
5. Crew transfer to/from surface in the Lander.
6. Crew return to Earth in Transfer Vehicle.
7. Transfer Vehicle aerobrakes and crew either return to Freedom or to Earth.



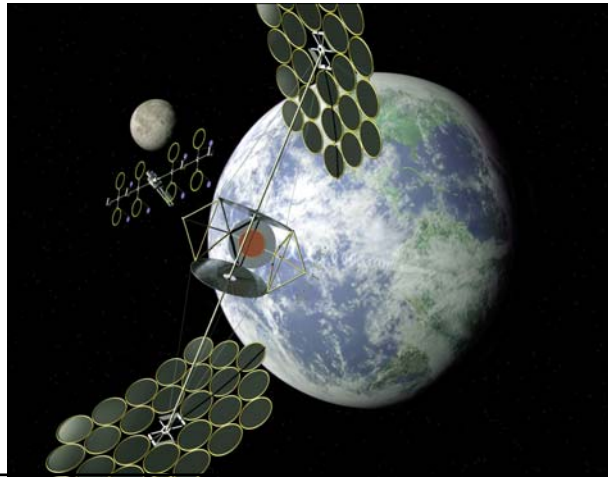
Advantages of Space Processing

Orbital Advantages

- Lots of Sunlight for power
- Zero Gravity
- Vacuum
- Transportation costs lower

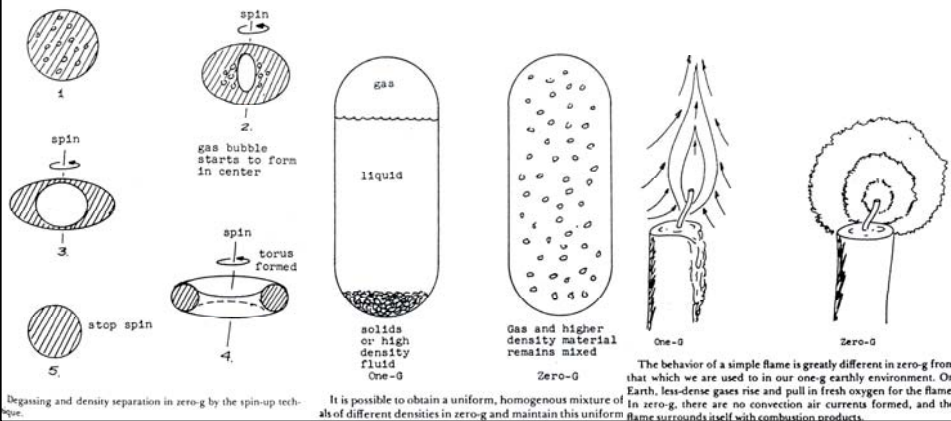
Orbital Problems

- Cooling: radiation only
- Stability



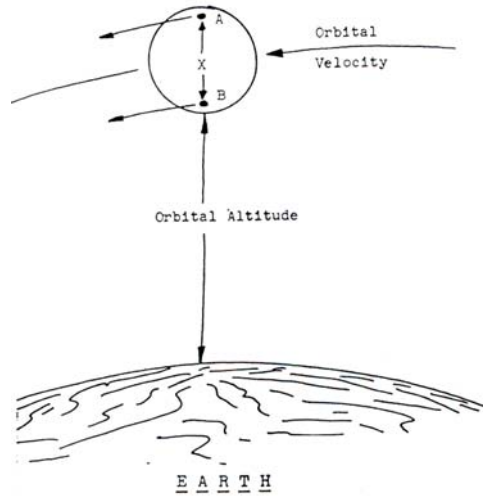
Microgravity

- Allows us to do many things not possible on earth
- Convection absent so create materials not possible on earth
- Make perfect spheres:
- Spin molten metal to remove gas: surface tension creates perfect sphere
- No Convection: Flame Burns in interesting ways



Microgravity Problems

- Not perfect in orbit due to Tidal Effect
- Problem with shaking of structure



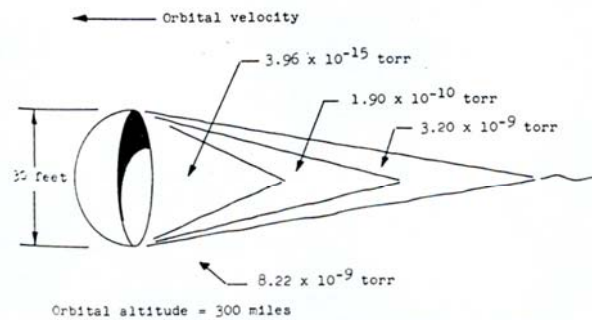
Potential Orbital Space Processes: Vacuum

Historical: Microgravity Material Processing

- Growth of high quality crystals
- Superior separation of materials
- Near perfect spheres (ball bearings etc)

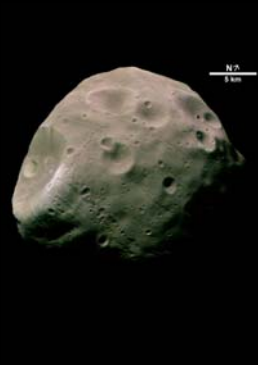
Newer Area: Vacuum Based Space Processing

- Ultra high vacuum for experimental work (Wakeshield)
- Microchip fabrication




Even though a very good vacuum exists at orbital altitudes, even higher vacuums exist in the wake of a satellite.





Raw Materials from Space High Value Metals in Asteroid/Meteors



Metal	LL Chondrite	90 th % Ni/Fe	98 th % Ni/Fe
Germanium	1020	70	35
Gallium*	N/A	N/A	87
Platinum	30.9	28.8	63.8
Ruthenium	22.2	20.7	45.9
Rhodium	4.2	3.9	8.6
Palladium	17.5	2.6	1.2
Osmium	15.2	14.1	31.3
Iridium	15.0	14.0	31.0
Gold	4.4	0.7	0.6

Table 13.1: PGM and Other Valuable Metals Concentrations in Grams Per Tonⁱⁱ

Lunar or Asteroid Profits

<i>Resource</i>	<i>Price 2004 (Kg)</i>	<i>Value Per 1 Million Kg Today's mkt</i>	<i>Lunar Discounted</i>	<i>Value Per 1 Million Kg</i>	<i>Percent Global Supply/yr</i>
Platinum	\$20,811	\$20.81 billion	\$10,405 0.3 x	\$10.4 billion	5X
Palladium	\$10,226	\$10.23 billion	\$2556 0.25 x	\$2.56 billion	5X
Ruthenium	\$1961	\$1.96 billion	\$490 0.25 x	\$490 million	12X
Rhodium	\$26,464	\$26.46 billion	\$6616 0.25 x	\$6.62 billion	15 X
Gold	\$14,109	\$14. 11 billion	\$3527 0.25 x	\$3.53 billion	.19 X
Iridium	N/A		N/A		
Osmium	N/A		N/A		
Germanium	\$470	\$470 million	\$117.5	\$117 million	22 X
Gallium	\$530	\$530 million	\$132	\$132 million	18 X
Total		\$74.57 billion		\$23.85 billion	

Table 13.2: USGS Metal Prices, Lunar Discount and Total Export Value

The New Race for Space

- USA
- Russia
- China
- Europe
- India
- Commercial

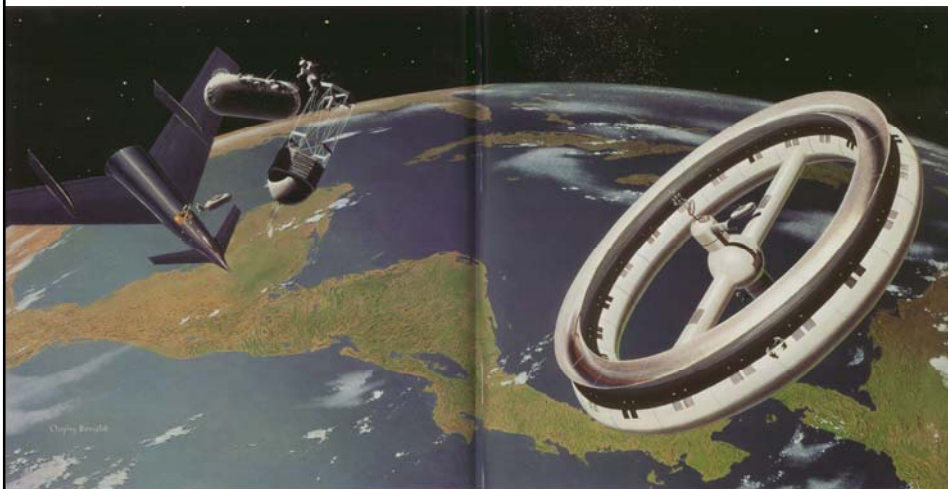


Reliable Access to Orbit
Von Braun's Shuttle
1952

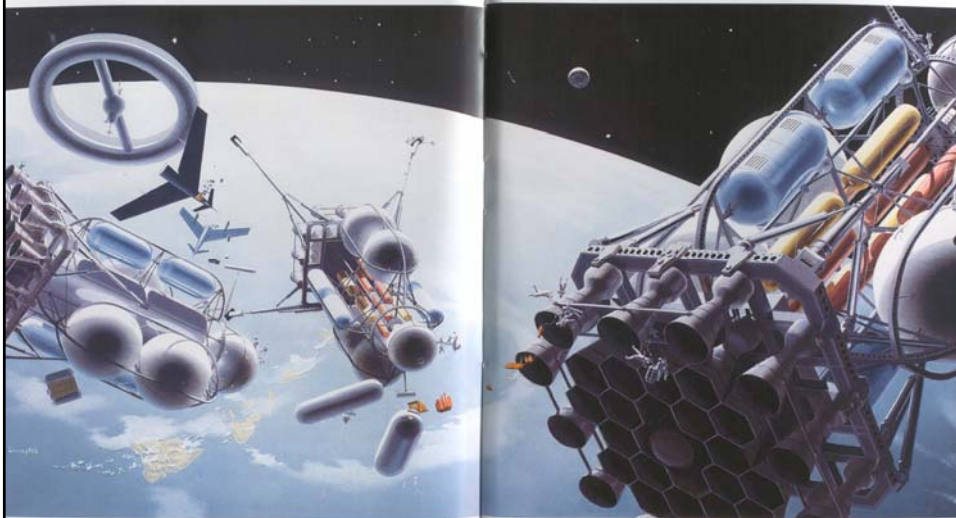
Science
Separation of the third stage of the manned ferry rocket 40 miles (64km) above the Pacific Ocean. From across the Space Frontier (1952). This was the first of the Collier's paintings, on which Rossmore and Winkler von Brauns collaborated. Von Braun asks Chrysler to look for showing the second-stage engines glowing red-hot, saying that a good engineer would have provided that by providing an adequate cooling system. (M on board, 18 x 20in (45 x 51cm).



Space Station Assembled in Low Earth Orbit
Von Braun's 1952 station

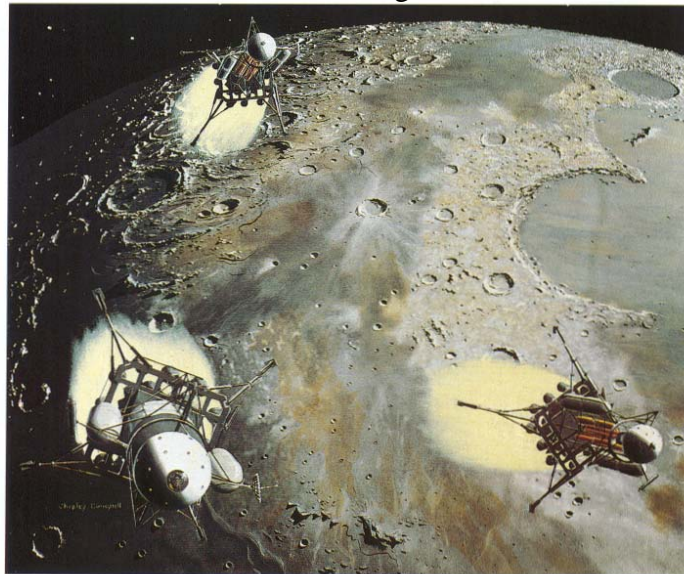


Moon Ships Built & Fueled at Space Station
Von Braun's 1952 design



Moon Ships Returning from the Moon to Space Station
Von Braun's 1952 design

RIGHT:
Landing on the Moon. From
Conquest of the Moon (1953).
In order to paint the Moon
correctly Chesley had a
sculptor, Nishan Tour, create
a 20in (51cm) lunar globe,
which he photographed. He
then enlarged and painted
over the photograph. Finally
the three ships were painted
on top of this.
Oil on board, 22 x 36 1/2 in
(56 x 93cm).



Mars Ships
Built at Station
Von Braun 1954



International Space Station



SPACE
C O M

International Space Station Final Plans



Image Courtesy of NASA

US Space Shuttle



US Shuttle Landing



US Atlas



Advance Soyuz Launcher



Proton Launcher

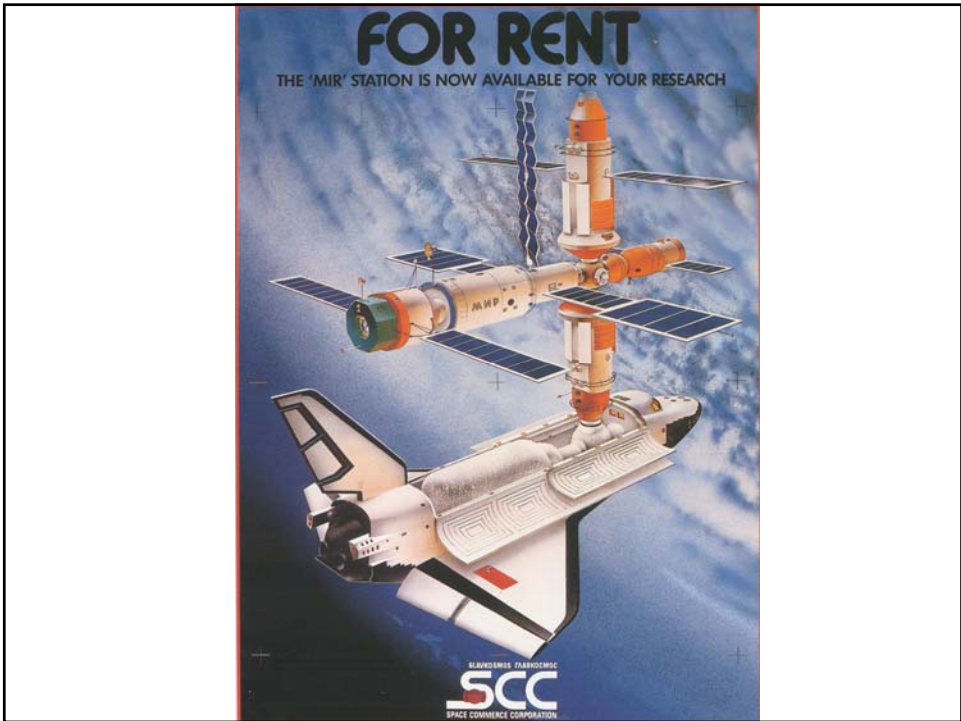
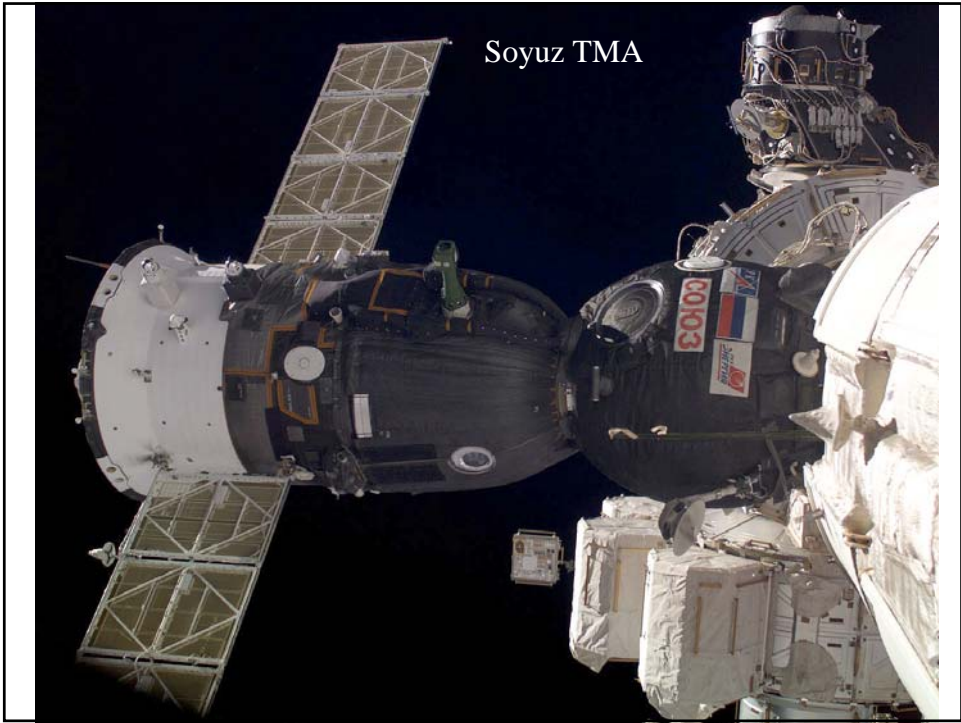


Zenit Booster

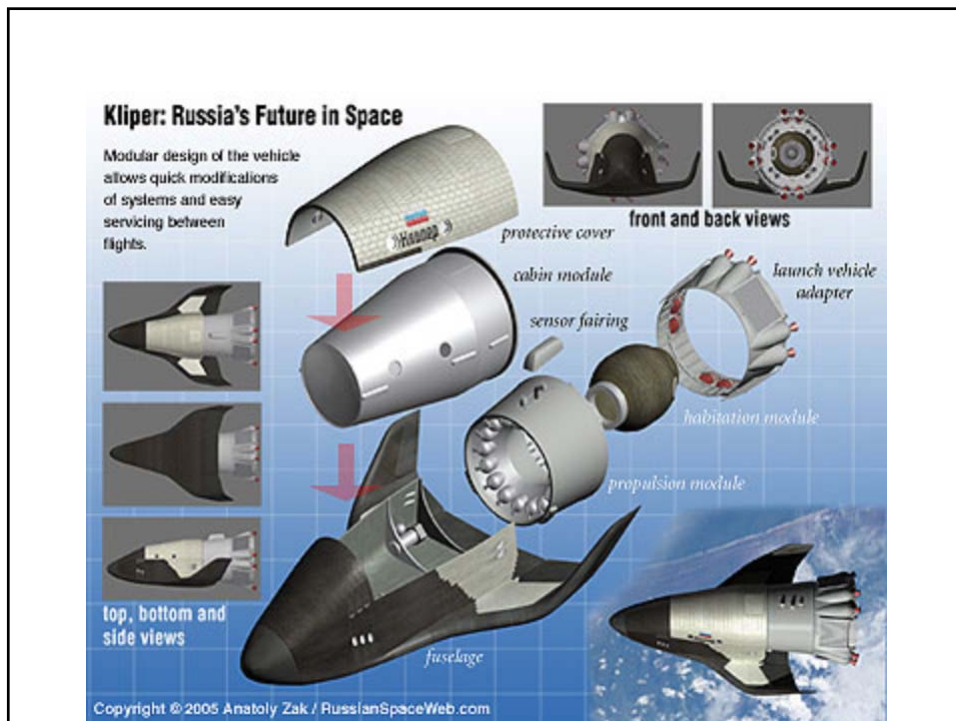


USSR's Buran Shuttle
Energia Booster

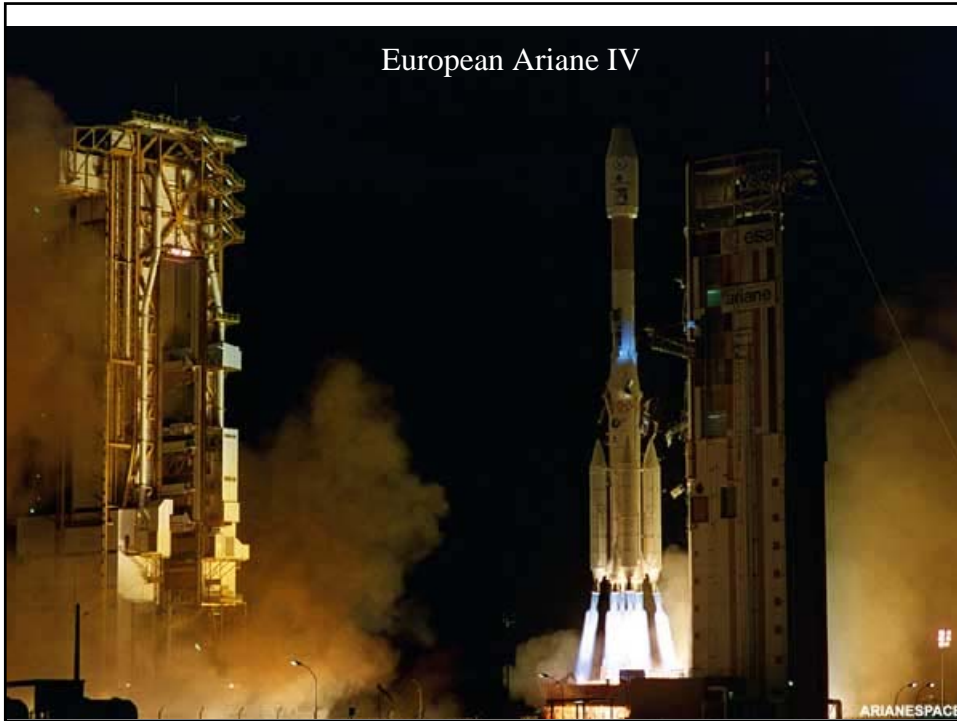




Russian Kliper Return Vehicle



European Ariane IV



European Ariane V



Japanese H2 (11T to orbit)



Indian GLSV: 10T to orbit



China's Long March Boosters



Shenzhou 5
Oct. 2003

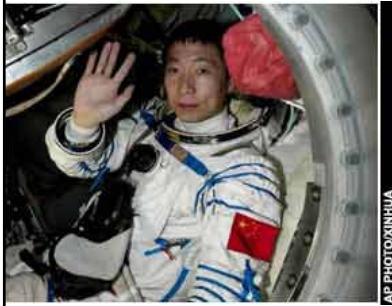


Shenzhou



Graphic © Simon Zajc

China's Taikonauts



Yang Liwei Shenzhou 5 2003

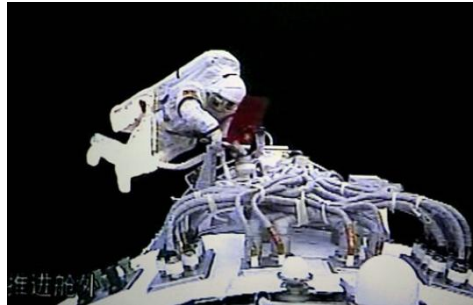


Shenzhou 6
Oct. 11, 2005
Fei Junlong &
Nie Haisheng

Shenzhou 7, Sept. 25, 2008



Zhiang, Boming, Haipeng



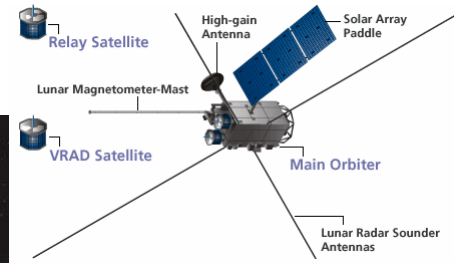
Zhiang 1st spacewalk Sept 27

Zhiang, Boming Spacewalk



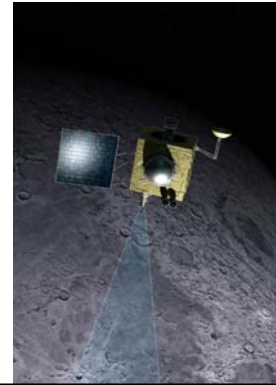
Japan & China's Lunar Orbiters (2007)

Japan's Selene (Kaguya)
HDTV camera on orbit (main
orbiter & 2 probes)



China's Chang'e 1 Nov
2007

India's Chandrayaan-1
Launched Oct. 23, 2008
Lunar orbit Nov 8
Moon Impact Probe Nov 14



Cheap Access to Space

- Requires a fully reusable space ship
- Fuel costs are low
- Most cost in building booster
- Throwaway Rockets: like destroying aircraft after each flight
- Boeing 747 costs \$200 million, 400 passengers
- Current cost of ticket to Japan \$1500
- If destroy plane after flight: \$500,000
- Cost about \$5000 /kg
- Orbital costs: about \$2000-\$10,000 / kg

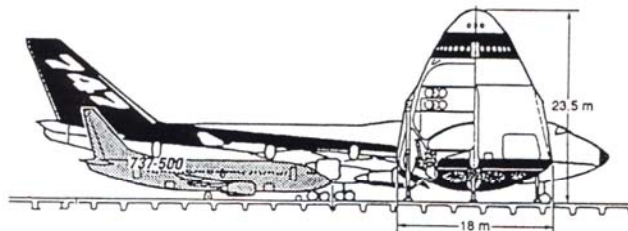
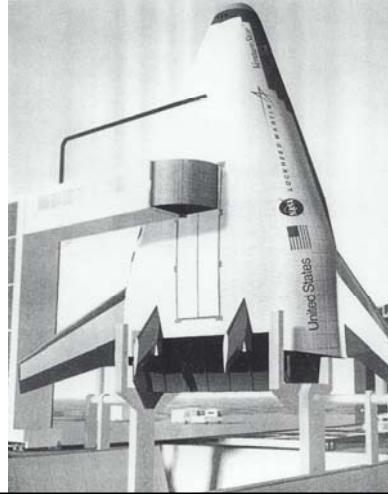


FIGURE 24-3: The proposed Japanese "Kankoh-maru" passenger-carrying SSTO compared in size to a Boeing 747-400 and a Boeing 737-500. (From Nagatomo et al, paper before 6th International Space Conference on Pacific-Basin Societies, December 1995.)

Single Stage To Orbit (SSTO)

- Launch & reuse like aircraft
- Use High Specific Impulse Liquid oxygen/Hydrogen fuels
- Must use very light airframe
- Advanced materials Carbon fiber/epoxy
- Reuseable reentry shield



50 Minutes from Anywhere

- Hypersonic vehicles could cover globe in 50 minutes
- Current half world flights 18 hours!
- Orient Express planes for free with SSTO or orbital vehicle

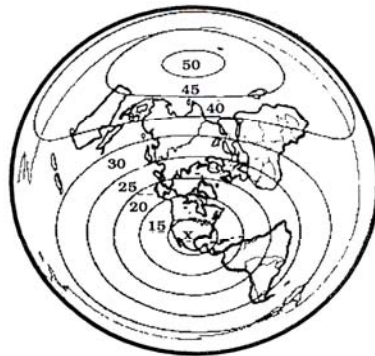


FIGURE 23-1: A polar projection of the world centered on a hypothetical Southwest Regional Spaceport in the United States. Circular areas containing numbers indicate the number of minutes of flight time required to fly there from the Southwest Regional Spaceport with Global Express. (Drawing from an SDIO/BMDO briefing paper.)

It's Economics Time

- Can a SSTO pay for itself?
- Even with high capital for 1 vehicle \$1-2 Billion
- Fuel costs have little effect
- Assume 100 flights per year, 5 year lifetime
- Cost per flight \$1.5 - 5 million
- Current Expendable costs \$50-\$700 million
- SSTO Lower by 10-100 times!
- SSTO cost per Kg \$1000 - \$100
- Problem is the high development cost

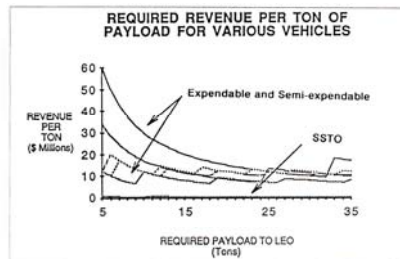
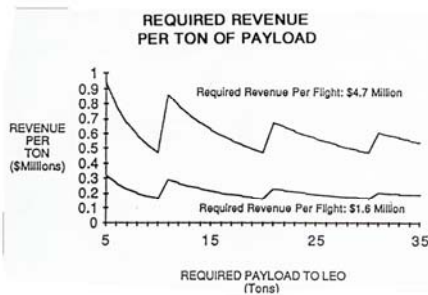


FIGURE 21-4: When the required revenue per ton and required payload to orbit are compared against existing expendable and semi-expendable (the space shuttle) vehicles, the SSTO has such drastically lower costs that they can hardly be seen on this chart. (Courtesy Paul C. Hans and The Enterprise Institute, Inc.)

Mass Fraction: Problem with Single Stage to Orbit

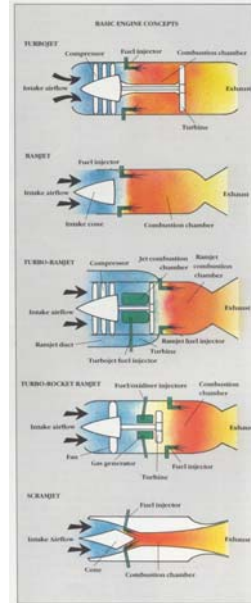
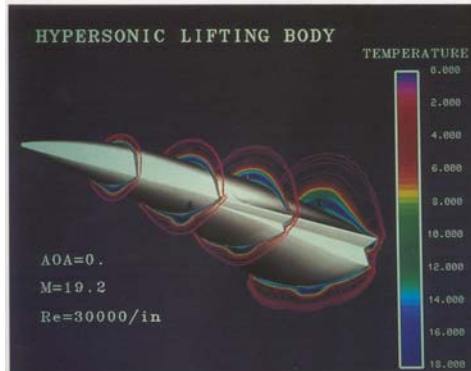
- Mass of rocket must be low compared to fuel
- Want mass fraction fuel/launcher nearly one
- Lower the mass fraction, poorer the launcher
- Typical 0.9-0.94

(Mass fraction = Propellant Weight/Gross Weight)

Vehicle	Propellant Weight (lb)	Gross Weight (lb)	Mass Fraction
Titan II Stage 1	260.0	269.0	0.966
Black Arrow Stage 1	28.7	31.1	0.922
Saturn V Stage 1	4584.0	4872.0	0.941
Titan III Stage 1	294.0	310.0	0.948
Titan IV Stage 1	340.0	359.0	0.947
Delta 6925 Stage 1	211.3	223.8	0.944
Atlas E	248.8	266.7	0.933
Saturn V Stage 2	993.0	1071.0	0.927
Zenith Stage 1	703.0	778.0	0.903
Titan III Stage 2	77.2	83.6	0.923
Saturn Ib Stage 2	233.0	255.0	0.913
Titan II Stage 2	59.0	65.0	0.908
Saturn Ib Stage 1	889.0	980.0	0.907
Ariane 5 Stage 1	342.0	375.0	0.912
Saturn V Stage 3	238.0	263.0	0.905
Energia core	1810.0	1995.0	0.907

Aircraft/Rocket combinations

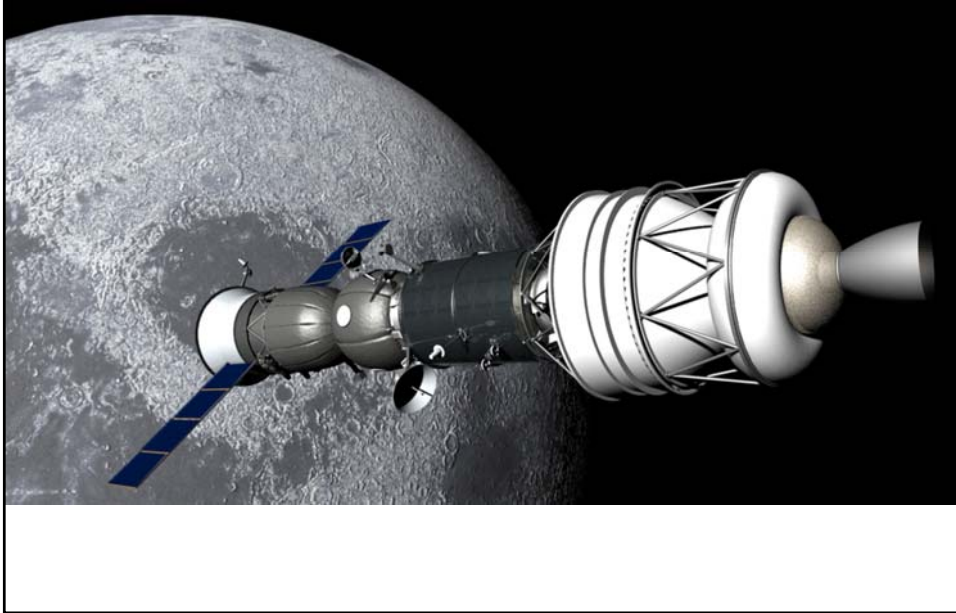
- Use air for oxidizer when possible
- Problem: need Supersonic Ram Jet engines
- work only Mach 3 - 6
- Jet engines before, Rockets after
- Use Hydrogen fuel
- Aircraft body must be integral with engines!



Economic Driver to Orbit: Space Tourism First by Dennis Tito May 16, 2001



Russian Soyuz Tourist Trip Around the Moon



Russian Soyuz Tourist Mission Around the Moon



X Prize: Driving Commercial Space Development



CREDIT: X PRIZE

Space Ship One & White Knight Carrier



Scaled Composites

White Knight 2 Shown 2008



NASA's Back to the Future: New Apollo Like Capsule



New Apollo Like Capsule



Apollo/LEM Lunar orbit flight



Larger/longer stay LEM



Same return capsule but on land

